

Control and perception techniques for aerial robotics

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Abstract

This paper review methods and technologies that have been applied in aerial robotics. The paper presents several unmanned aerial vehicle platforms. Then summarizes different control techniques including both control architectures and control methods. Furthermore, computer vision techniques for aerial robotics are briefly considered. Finally, the paper presents systems and projects involving multiple autonomous aerial and ground systems.

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1. Introduction

In the last decades many autonomous and teleoperated vehicles for field robotics applications have been developed, including wheeled, tracked and legged vehicles. However, in many cases, ground vehicles have significant inherent limitations to access to the desired locations due to the characteristics of the terrain and the presence of obstacles that cannot be avoided. In these cases aerial vehicles is the natural way to approach the objective to get information or even to perform some actions such as the deployment of instrumentation. Then, aerial robotics seems a useful approach to perform tasks such as data and image acquisition of targets and affected areas, localization of targets, tracking, map building and others.

Unmanned aerial vehicles (UAV) have been used for military applications but also are useful for many civilian applications such as terrain and utilities inspection, disaster monitoring, environmental surveillance, search and rescue, law enforcement, aerial mapping, traffic surveillance, and cinematography. In the last years UAVs improved their autonomy both in energy and information processing. However, the development of autonomous aerial robotic

vehicles involves many problems related to limited payload, safety requirements, flight endurance and others.

This paper reviews some significant developments in aerial robotics. In Section 2 different UAV platforms are reviewed. Section 3 is devoted to UAV control emphasizing autonomous helicopter control. The environment perception techniques are considered in Section 4. Section 5 is devoted to multi-robot systems including a short presentation of the COMETS project on multiple heterogeneous aerial vehicles. The paper closes with the conclusions and the references.

2. UAV platforms

Unmanned air vehicles (UAVs) are self-propelled air vehicles that are either remotely controlled or are capable of conducting autonomous operations. UAV experimental research ranges from low-level flight control algorithm design to high level multiple aircraft coordination.

During the last decades significant efforts have been devoted to increase the flight endurance and payload of UAVs. Thus, there are High Altitude Long Endurance (HALE) UAVs, as for example the Northrop Grumman Ryan's Global Hawks (65000 ft altitude, 35 h flight and 1900 lbs payload) and Medium Altitude Long Endurance (MALE) UAVs, as for example the General Atomics'



Fig. 1. The Predator from General Atomics.

Predator (see Fig. 1, with 27,000 ft altitude, 30/40 h flight and 450 lbs payload), and the Tactical UAVs such as the Pioneer with 15,000 ft altitude, 5–6 h flight and 25 kg payload. In the last years man portable or hand launched UAVs, called “Organics UAV”, such as Pointer (AeroVironment), Javelin (BAI) or Black Pack Mini (Mission’s Technologies) have been presented.

Furthermore, many different Vertical Take-Off and Landing (VTOL) UAVs including helicopters and several designs such as the Guardian from Bombardier, and the Sikorsky’s Cypher or Dragon Warrior which can be operated in either wings-on (see Fig. 2) or wings-off configurations.

On the other hand, in the last years, Micro Air Vehicles, with dimensions lower than 15 cm, have gained a lot of attention. These include the Black Widow manufactured by AeroVironment (see Fig. 3), the MicroStar from BAE and many new designs and concepts presented in several Universities such as Entomopter (Georgia Institute of Technology), Micro Bat (California Institute of Technology), MFI (Berkeley University), as well as other designs in European Research Centres.

In the *Unmanned Aerial Vehicles Roadmap (2001)* a survey of platforms and UAV technologies is presented.

In many aerial robotic projects, adaptation of conventional remote controlled model aircrafts with modest flight endurance, altitude and payload are used. Some of these platforms are the same than used for entertainment and applications such as aerial photography, cinematography, chemical spraying, inspection and other tasks in which the



Fig. 2. The Sikorsky’s Dragon Warrior (Cypher2).



Fig. 3. Black widow (AeroVironment).

vehicles are maintained in the line of sight of the human pilot. In some cases the evolution of the conventional platforms led to new vehicles with increased flight endurance and payload such as the Yamaha R50 and Rmax. Thus, the Rmax is able to fly for 1 h carrying a 24-kg payload.

Aerial robotics has mainly involved helicopters and other VTOL designs, airships and fixed wing small UAVs. The main advantage of helicopters and other VTOL platforms is the manoeuvrability, which is needed for many robotic applications. The ability to maintain the aerial vehicle in hovering is very important in many tasks. However, they are difficult to control and require experienced safety pilots for their development and application. Moreover, fully autonomous control of helicopters is a difficult task that requires the application of reliable and nonlinear control laws.

Several Universities in the USA have developed autonomous helicopters. Thus, the Robotics Institute at Carnegie Mellon University (CMU) conducted since the early nineties an autonomous helicopter project. They have developed different prototypes from small electrical radio controlled vehicles to autonomous helicopters using the Yamaha R50 platform. The autonomous CMU helicopter won the AUVSI aerial robotic competition in 1997.

The University of Southern California (USC) conducted, since 1991, an autonomous helicopter project developing several prototypes, such as the Autonomous Vehicle Aerial Tracking and Retrieval/Reconnaissance (AVATAR) prototypes presented in 1994 and 1997. The AVATAR helicopter won the AUVSI Aerial Robotics Competition in 1994.

The University of Berkeley also developed autonomous helicopters in the Berkeley AeRobot project, BEAR, in which the autonomous aerial robot is a testbed for an integrated approach to intelligent systems.

The Georgia Institute of Technology has the Unmanned Aerial Vehicle Research facility and developed several platforms and aerial autonomous systems during the last decade. GIT also won the AUVSI Aerial Robotics Competition.



Fig. 4. Yamaha Rmax platform used in the WITAS project.

In Europe the University of Linköping is conducting the WITAS project which is a long-term basic research project involving cooperation with other Universities and private companies (Doherty et al., 2000). The Yamaha Rmax helicopter is currently being used in the WITAS project (see Fig. 4). Moreover, several Universities such as the Technical University of Berlin, ETH Zurich (Eck, Chapuis, & Geering, 2001), and Universidad Politécnica de Madrid (Del-Cerro, Barrientos, Campoy, & García, 2002) are using the adaptation of conventional radio controlled helicopters with different autonomous capabilities.

Fig. 5 shows MARVIN developed by the Technical University of Berlin (Remuss, Musial, & Hommel, 2002), which won the AUVSI Aerial Robotics Competition in 2000, and Fig. 6 the helicopter being developed jointly by the University of Seville and Helivision.

Airships (blimps) are stable platforms to take images and in case of failure present a graceful degradation which is not the case of other platforms such as helicopters. Furthermore, they can be piloted without extensive training. However, airships have significant manoeuvrability constraints, they are larger and the deployment is more difficult. Moreover, they can only fly when the wind velocity is low. Airship platforms are also used in aerial robotics projects, such as



Fig. 5. The MARVIN autonomous helicopter flying in experiments of the COMETS project (May 2003).

3. UAV control

3.1. Control architectures

On-board control architectures for UAV have to integrate a variety of sensor information (GPS, three-axis rate gyro, three-axis accelerometer, aircraft attitude reference sensor, compass, altitude sensors among others), and low level motion servo-controllers, to control the vehicle typically in different control modes. Eventually, environment perception, object tracking and local reactive (obstacle avoidance) and planning capabilities are considered. However, in existing aerial robotic prototypes these capabilities are modest when comparing to

Fig. 6. University of Seville-Helivision helicopter flying in experiments of the COMETS project (May 2003).

Karma (see Fig. 7) at LAAS (CNRS, France), Autonomous Unmanned Remote Monitoring Robotic Airship (AUR-ORA) at CENPRA (Brazil) (Bueno et al., 2002), and the airship of the University of Stuttgart (Wimmer et al., 2002).

Conventional fixed wing airplanes also have manoeuvrability constraints. The lack of hovering capabilities imposes significant limitations for their application in aerial robotics. However, the reach and flight endurance can be larger than helicopters and other VTOL designs. Furthermore, both manual and autonomous control are simpler. Then, many autonomous airplanes have been developed and used for reconnaissance, surveillance, environment monitoring and others. Some of these platforms are also used in aerial robotics projects involving localisation and mapping functions such as the delta wing Unmanned Aerial Vehicle Brumby at the University of Sydney, which has been designed to fly in excess of 100 knots and currently has an endurance of 1/2 to 1 h flight time. The aircraft has the capacity to carry up to 6-kg payload when remotely piloted, or 4 kg when operated autonomously.



Fig. 7. Karma, developed at LAAS-CNRS, flying in experiments of the COMETS project (May 2003).

ground robots. In fact, the on-board hardware is seriously constrained by the load and energy consumption. On-board UAV control hardware is an ideal application for new embedded control systems involving FPGAs, DSPs and new powerful microcontrollers. However, other hardware platforms such as the PC-104 computer system with real-time Operating System are also applied to simplify the development.

The on-board control hardware is linked to an operator ground controller which is used to send commands and GPS corrections to the on-board controller and to visualize information transmitted from the UAV. In many projects these controllers are now implemented by means of laptops.

The control of fixed wing aircrafts can be considered at different levels (McLean, 1990). Low-level control is called stability augmentation in the airplane control domain. The objectives are perturbation rejection and improving the dynamic response when the pilot or the higher level controller/guidance system provides commands. Fig. 8 shows some of the involved variables. The linear (u, v, w) and angular velocities (p, q, r) in the airplane body axis, and orientation angles are usually considered as state variables. The control variables are the deflections of aerodynamic control surfaces (ailerons, elevator and rudder) about the three-body axis and the propulsion source (throttle). The airplane dynamics, and then the control problem, is usually decoupled in longitudinal (pitching rate q control using elevator) and lateral/directional control where the rolling

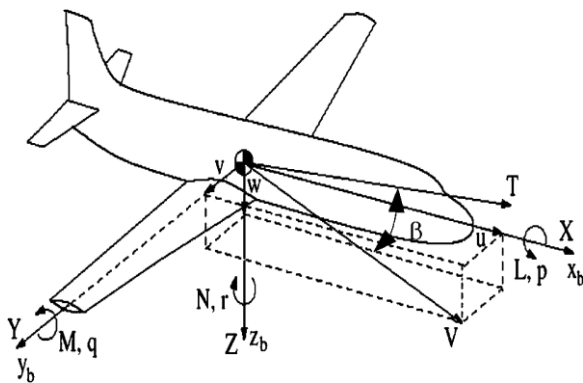


Fig. 8. Variables for UAV control.

rate (p) and the yawing rate (r) are controlled by means of the aileron and rudder.

The airplane attitude control consist of functions that allow the aircraft to be placed, and maintained, in any required orientation in space, stability augmentation being the inner control loop (McLean, 1990). The variables to be controlled are often the pitch and yaw angles, and the airplane side slip angle β . Furthermore, the objective of this control level is to perform coordinated turns without lateral acceleration and airplane sideslip. On the top of the control hierarchy is the flight path or trajectory tracking (which includes precision timing). In this case the course angle error and the cross track distance can be considered as error signals in the guidance

hierarchical architecture segments the control tasks into different layer of abstraction in which planning, interaction with the environment and control activities are involved. The hierarchical flight management system (Kim & Shim, 2003) has a stabilization/tracking layer, a trajectory generation layer, responsible for generating a desired trajectory or a sequence of flight modes, and a switching layer which switches between several strategy planners. Both continuous and discrete event systems are considered. In order to model these control systems, hybrid system theory has been proposed (see for example Koo, Hoffman, Shim, Sinopoli, & Sastry, 1998).

loop. The trajectory to be tracked can be provided by a trajectory generator which is designed to interpolate between waypoints or target points to achieve a given mission. Trajectory tracking and generation can be jointly considered by using approaches in which the dynamics of the vehicle is also used for trajectory generation and the target turning rates and heading angles obtained by the trajectory generator are used for control.

Several commercial autopilots for remotely piloted airplanes that are able to follow way points are in the market. Typically Kalman filtering techniques are applied for position estimation and the reliability and accuracy depend on the position measurement technology applied. However, the autonomous control of helicopters and other VTOL platforms with different control modes is more complex and motivated the research activities of several universities.

In the following, this section concentrates in autonomous helicopter control. The position and orientation of an helicopter (six degrees of freedom) are usually controlled by means of five control inputs: the main rotor thrust (power to the rotor), the collective input (main rotor blade angle) which has a direct effect on the helicopter height (altitude control), the tail rotor which affects the heading of the helicopter (yaw motion) and compensates the anti-torque generated by the main rotor, the longitudinal cyclic which modifies the helicopter pitch angle and the longitudinal translation, and the lateral cyclic, which affects the helicopter roll angle and the lateral translation (lateral cyclic). Then, it is a multivariable nonlinear underactuated system with strong coupling in some control loops.

The USC developed a behaviour-based architecture for the control of the AVATAR helicopter (Fagg, Lewis, Montgomery, & Bekey, 1993). The low-level behaviours correspond to the generation of the four input commands of the helicopter (collective throttle, tail rotor, longitudinal and lateral cyclic). The second level implements short-term goal behaviours: transition to altitude and lateral velocity. The highest-level behaviour, navigation control, is responsible for long-term goals such as moving to a particular position and heading.

Intelligent control architectures for unmanned air vehicles (helicopters) are also researched at Berkeley. The

GTI also developed autonomous helicopter control systems and research in flight controls, avionics and software systems.

3.2. Learning and pilot knowledge-based control methods

As far as the research in control methods is concerned different approaches can be used. Thus, fuzzy logic has been applied to control the Yamaha's helicopter at the Tokyo Institute of Technology, which demonstrated autonomous capabilities and also person-machine interfaces including voice command (Sugeno, Griffin, & Bastian, 1993).

Fuzzy logic with rules generated by the observation of a human pilot and consultation with helicopter experts is the approach used in Cavalcante and coworkers (1995).

In Montgomery, Fagg, and Bekey (1995) the behaviours of the control architecture proposed in the USC architecture are implemented as PD control loops with gains tuned by trial and error. In Montgomery and Bekey (1998), the “teaching by showing” approach is presented. In this work the controller is generated by using training data gathered while a human teacher controls a system until the synthesized controller can also control the system to meet predefined performance criteria.

In Buskey, Wyeth, and Roberts (2001) learning is based on the direct mapping of sensor inputs to actuator control via an artificial neural network. Then, the neural network controller was used for the helicopter hovering.

The analysis of the pilot’s execution of aggressive manoeuvres from flight test data is the base of the method presented in Gavrillets, Frazzoli, Mettler, Piedmonte, and Feron (2001) to develop a full non-linear dynamic model of a helicopter. This model will be used in the design of new control systems for autonomous helicopters.

3.3. Model-based control methods

On the other hand, several methods have been applied for model-based control of UAVs. Modelling the UAV dynamics is a main issue. The full model of a helicopter involving the flexibility of the rotors and fuselage and the dynamics of the actuators and the combustion engine is very complex. Then, in most cases, the helicopter is considered as a rigid body with inputs forces and torques applied to the

centre of mass and outputs the position and linear velocities of the centre of mass, as well as the rotation angles and angular velocities. Furthermore, the relations between the control inputs of the helicopter and the above mentioned forces and torques should be considered in the model. In general, these relations involve the consideration of the aerodynamics of the fuselage and the effect of stabilizers. However, at low speeds these effects can be ignored (Koo & Sastry, 1998).

In Kim and Tilbury (2004) a mathematical model and experimental identification of a model helicopter is presented. The model of the interactions between the stabilizer flybar and the main rotor blade is also included showing its effects in the stability of the model helicopter. The identification of the parameters is performed on a SISO basis using a specially built stands to restrict the motion of the helicopter to one degree of freedom. It should be noted that the identification from input–output data, collected when a human pilot is controlling the vehicle, is difficult because it is not possible to study the individual effect of each control input (the pilot has to apply more than one input to maintain the stability).

In Mettler, Tischler, and Kanade (2002) a parametrized model of the Yamaha R-50 autonomous helicopter is identified using frequency domain methods. The stabilizer bar

is also taken into account. The model is verified with special flight experiments using doublet-like control inputs in hover and forward flight showing its ability to predict the time domain response of the helicopter to control inputs.

It has been shown that the multivariable nonlinear helicopter model cannot be converted into a controllable linear system via exact state space linearization. In addition, for certain output functions, exact input–output linearization results in unstable zero dynamics. However, if only the position and heading are chosen as outputs, by neglecting the coupling between moment and forces, the approximated system with dynamic decoupling is full state linearizable and output tracking can be applied (Koo & Sastry, 1998).

It should be noted that in hovering, the nonlinear system can be linearized and then multivariable linear control techniques such as LQR and H^1 can be applied. In Kim and Shim (2003) multi-loop linear PID techniques also obtained good results when applied to the Yamaha R-50. However, if large perturbations should be compensated, or significant tracking abilities are required, this strategy could not be enough. In this case further improvements can be obtained by adding nonlinear control terms that compensate significant deviations with respect to the hovering conditions.

In Kadmiry, Bergsten, and Driankov (2001) a fuzzy gain-scheduling approach, based on the linearization of the original nonlinear helicopter model, is proposed and tested in simulation.

Johnson and Kannan (2002) combine the helicopter attitude inner control loop and the outer trajectory control loop and apply adaptive techniques to cancel model errors by preventing unwanted adaptation to actuator limits and dynamics in the inner loop.

In Shim, Koo, Hoffman, and Sastry (1998) linear robust multivariable control, fuzzy logic control and nonlinear tracking control are compared in the simulation of two scenarios: vertical climb and simultaneous longitudinal and lateral motion. It is noted that nonlinear control techniques by applying feedback linearization are more general and cover wider ranges of flight envelopes but requires accurate knowledge about the system and are sensitive to model disparities, such as changes in the payload, or to the aerodynamic thrust–torque model.

In general no guarantee of robustness against model uncertainties or disturbances and no adaptive capabilities are provided by many feedback linearization techniques. However, in some cases, nonlinear controller robustness properties are increased using sliding mode and Lyapunov-based control (Maharaj, 1994). Typically, these techniques trade the controller performance against uncertainty, but require a priori estimates of parameter bounds, which may be difficult to obtain.

However, research efforts to design new robust non linear control laws are pursued. Then in Isidori, Marconi, and Serrani (2001) the vertical motion of a nonlinear model of a helicopter to a reference signal, while stabilizing the lateral and longitudinal position and maintaining a constant attitude, is studied. The problem is motivated by the

synchronization of the vertical motion of the helicopter with a sea carrier subject to wave-induced oscillations, and then the reference signals are sum of sinusoidal signals (assumed not to be available to the controller). A nonlinear adaptive output regulation and robust stabilization of systems in feed-forward form by means of saturated control is applied in simulation. The simulation results show robustness against uncertainties on the model and on the exogenous reference signal. The method also requires the a priori computation of robustness bounds.

In Kim and Shim (2003), the application of nonlinear model predictive control is proposed. At each sample time, the controller computes a finite control sequence which minimizes a quadratic index. This index includes the errors of the outputs (helicopter coordinates and heading) with respect to desired trajectories, additional state variables which should be bounded, and the control actions (longitudinal and lateral cyclic pitch, collective, and tail rotor). A gradient descent technique is used to compute the optimal values of the control variable. The method improves the tracking performance at the expenses of heavy computing load.

In Fantoni and Lozano (2002) the control of under-actuated systems including helicopters and Planar VTOL (PVTOL) is studied. Several control techniques are presented including backstepping, energy-based controllers and Lyapunov-based controllers.

At CMU a high-order linear model of the R-50 Yamaha helicopter is used for control. This model was extracted by

using the Modelling for Flight Simulation and Control Analysis (MOSCA) with a nonlinear simulation model of the helicopter. The controller consists of one multivariable (MIMO) inner loop for stabilization and four separate (SISO) guidance loops for velocity and position control. Several manoeuvre tests have been conducted with the helicopter (square, forward turn, backward turn and nose-out circle). The controller is designed for hovering but its robustness leads the helicopter to perform the manoeuvres efficiently even if the trajectories are not optimal (La Civita, Papageorgiou, Messner, & Kanade, 2003).

4. Environment perception techniques

determining the position and orientation of an aircraft with respect to a ship from the images of several points taken by an infrared camera.

In the BEAR project, vision-based pose-estimation of unmanned helicopters relative to a landing target and vision-based landing of an aerial vehicle on a moving deck are researched (Shakernia, Vidal, Sharp, Ma, & Sastry, 2002; Vidal, Sastry, Kim, Shakernia, & Shim, 2002). In this project, results on multiple view geometry are used to impose constraints over multiple views of a planar landing target with a single camera. These constraints, summarized in the so called multiple view matrix, allows to compute the displacement of the UAV given three or more views of the target. The known size of the landing target allows to obtain

Environment perception technologies used within UAVs includes cameras and range sensors. For some particular operations such as autonomous landing, range sensors (laser and ultrasonics) are widely used. However, computer vision plays the most important role. Computer vision is used for several applications, from autonomous UAV control to detection, monitoring or terrain mapping.

4.1. Vision-based position estimation

Vision is used as a positioning sensor in several UAV projects. It is argued that some applications that need a very accurate relative position of the UAV respect to objects, as close proximity flight, can not rely on inertial navigation systems or global positioning systems (Amidi, 1996). Thus, vision is used as a method to sense relative position.

The concept of visual odometer (Amidi, Kanade, & Fujita, 1998) was implemented in the CMU autonomous helicopter. Using this concept the helicopter can visually lock-on to ground objects and sense relative helicopter position and velocity in real time. Template matching techniques are used to estimate the displacement of the object between consecutive images. This, combined with angular rate sensing, allows to estimate the UAV motion. In this case only a few templates are tracked. Each cycle new targets are selected to avoid losing targets. The same visual tracking techniques, combined with inertial sensors, have been applied to autonomously take off, follow a prescribed trajectory, and landing. The CMU autonomous helicopter also demonstrated autonomous tracking capabilities of moving objects by using only on-board hardware.

The use of vision for autonomous landing has been actively researched. In the early nineties, Dickmanns and Schell (1992) presented some results of the possible use of vision for landing an airplane. Zhang and Hintz (1995) developed a video-based attitude and height sensor for low altitude airborne vehicles. Neural networks are used to compute attitude and height from the measures obtained by a video camera that observes the structured light pattern created by a set of infrared lasers. Yakimenko, Kaminer, Lentz, and Ghyzel (2002) consider the problem of

the scale of the displacement. Using off-the-shelf hardware, the vision system in the BEAR project is able to operate at 30 Hz. The feature extraction and matching are simplified by using an artificial target of known shape. Vision is situated in the control loop as a high lever supervisor which sends the current position of the landing target to the navigation controller.

Computer vision is also used for safe landing in the AVATAR project. Thus, in Garcia-Pardo, Sukhatme, and Montgomery (2001) a strategy and an algorithm relying on image processing techniques to search the ground for a safe place to land is presented. In Saripally and Sukhatme (2003) a system for landing the AVATAR helicopter in a slow moving helipad of known shape is presented. The system applies target recognition using computer vision (through threshold-based segmentation and using the

moments of inertia of the known helipad) and a Kalman filter for target tracking.

Saripally and Sukhatme (2003) also present a technique for helicopter position estimation using a single CMOS pointing downward camera with a large field of view and a laser pointer to project a signature onto the surface below in such a way that can be easily distinguished from other features on the ground.

Most of previous techniques rely on feature matching techniques. Recent focus has been put on algorithms for extraction of robust and invariant features for wide-baseline matching. Thus, in the WITAS UAV project, a method for robust extraction of affinely invariant features has been developed (Forsee'n & Granlund, 2003).

4.2. Simultaneous localization and mapping (SLAM) with UAVs using vision

SLAM techniques are closely related to the position estimation problems. SLAM is now a well known technique and has been used mainly with ground robots. Among them, only a few are based mainly on vision sensors. Also, there are some applications of vision for SLAM with UAVs.

Terrain mapping techniques have been developed at the LAAS (Lacroix, Jung, & Mallet, 2001). The perception system being designed for the Karma airship (Lacroix et al.,

2001, Lacroix, Jung, Soueres, Hygounenc, & Berry, 2002) applies stereo vision, interest point matching and Kalman filtering techniques for simultaneous localization and mapping using only vision. The robust interest point matching algorithm combined with stereo vision is used for blimp position estimation. Some of the interest points are used as landmarks in the SLAM approach for the filter update phase. By using stereo vision a dense range map can be obtained. Karma carries a stereo bench 2 m wide (place along the blimp gondola), and is able to build maps with an accuracy of 10 cm height and a cell resolution of 5 cm². Fig. 9 shows an example of a map generated by the blimp Karma.

UAV simultaneous localisation and map building with vision using a delta fixed wing platform is also presented in Kim and Sukkarieh (2003). Artificial landmarks of known size are used in order to simplify the landmark identification problem. The known size of the landmarks allows to use the cameras as a passive range/bearing/elevation sensor. In this work images of known landmarks and inertial data are used as ground truth and they show the corrective properties of the SLAM algorithm.

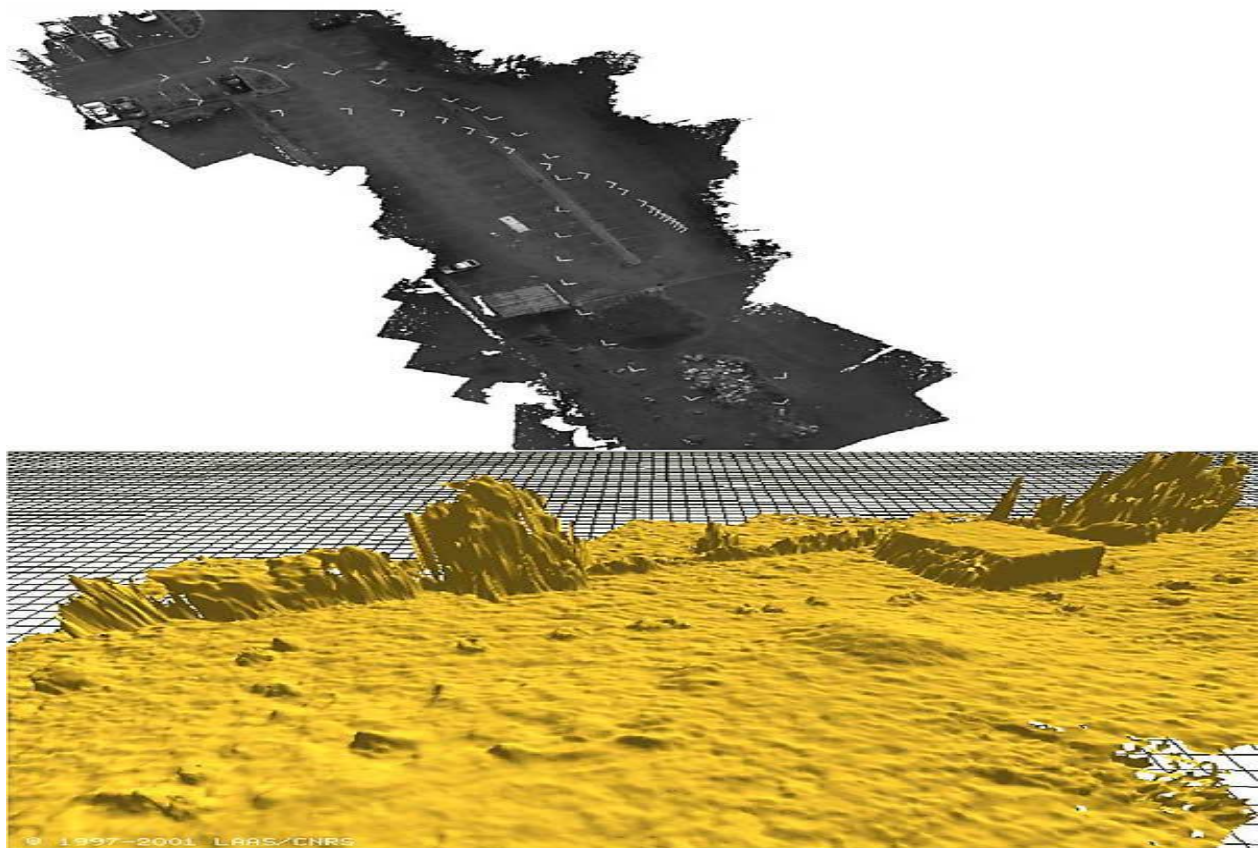


Fig. 9. Orthoimage and elevation map generated from a set of 100 aerial images taken by the blimp Karma.

Also, scanner laser sensors have been used in the CMU helicopters to build terrain maps. This helicopter built accurate aerial maps of the experimental site of the VSAM project and portions of the Houghton crater in Devon Island for NASA geologists.

4.3. Monitoring and other applications

The mobility of UAVs providing the possibility of changing at will, the point of view make them very suitable vehicles for monitoring applications.

Detecting moving objects on the image plane is an important function for monitoring applications. Thus, novel and fast algorithms for dense motion estimation have been developed within the WITAS project. In [Farneback and Nordberg \(2002\)](#), a two frame motion estimation algorithm based on a multi-scale polynomial expansion of the images allows to obtain a dense motion field, used for ground vehicle detection in traffic monitoring applications. Other computer vision activities in the WITAS project are described in [Nordberg, Doherty et al. \(2002\)](#).

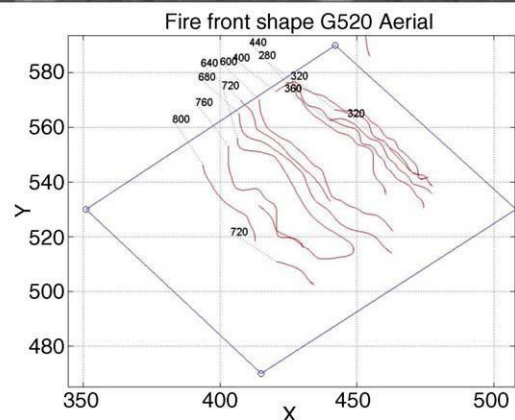
In the BEAR project, vision is also used to detect evaders and determine their position in pursuit-evasion games ([Vidal et al., 2002](#)). The evaders are detected looking for objects that move independently on the image plane. The number of independent moving objects is obtained analyzing the optical flow matrix for multiple points across several frames.

Motion estimation, object identification and geolocation by means of computer vision are also researched in the framework of the COMETS project ([Merino & Ollero, 2002a, 2002b](#)). In this project fire monitoring using UAVs is considered as an application scenario. The perception system designed in this project also implements image stabilization by using visual tracking techniques. Furthermore, the above mentioned terrain mapping techniques with the Karma airship are also being integrated in the COMETS system and will be used to provide terrain maps required for geolocation and other applications.

[Fig. 10](#) shows the result of using aerial images for fire monitoring. Threshold-based techniques are used for fire segmentation. The fire front is identified and then geolocated. Image stabilization is used to track correctly the position of the fire front. [Fig. 10](#) (bottom panel) shows the evolution of the fire front each 40 s for an experiment of controlled forest fires carried out in Portugal (May 2002).

Other applications of computer vision are the detection of targets and the interpretation of scenes from the aerial images. Thus, for example, the MARVIN helicopter at TUB and the CMU helicopter have used vision during the AUVSI Unmanned Aerial Robotic Competition to achieve the goals of the competition.

MARVIN ([Remuss, Musial & Hommel, 2002](#)) uses a still digital camera. This camera is able to take high-quality images of large resolution at low frame rates (640 480 at 0.2 Hz or 1024 768 at 0.08 Hz). Combined with an



[Fig. 10](#). Experiments of fire monitoring using aerial images. Top panel: one of the images taken. Bottom panel: estimated fire front evolution.

adequate searching strategy, they succeeded looking for casualties in the scenario of the competition.

CMU helicopter uses principal component analysis to store templates of interesting objects, and to detect them posteriorly. This has been used successfully for the detection of hazard labels on barrels during the AUVSI competition.

Object identification of ground targets by using previously known appearance or colour has been also implemented by several authors. In [Merino et al. \(2003\)](#), the identification of other aerial vehicles from on-board cameras by using a phase-only matched filter is presented. [Bueno et al. \(2002\)](#) present results on-road identification and tracking using vision in the AURORA airship. They use spectral characteristics of the targets for identification purposes.

Image processing techniques have been proposed for cable segmentation and tracking for inspection of power lines ([Del-Cerro et al., 2002](#)). Stereo vision is also proposed to maintain the UAV at a desired distance from the inspected cables.

4.4. Other perception sensors

Omnidirectional cameras have been used on ground robots for navigation and other purposes (i.e. formation

control in [Das et al., 2002](#)). Initial approaches for its use in UAVs are being presented. Thus, in [Hrabar and Sukhatme \(2003\)](#) an experiment to study the feasibility of using omnidirectional cameras within a VTOL UAV is presented. In the tests, they use the omnidirectional camera to detect a set of targets and to generate control commands to situate the UAV at the centroid of these targets.

In [Netter and Franceschini \(2002\)](#), an artificial camera eye, consisting of a one dimensional 20-pixel linear photoreceptor and analog electronic processing circuits are used for terrain following and obstacle avoidance in a test-bed for MAVs.

4.5. Implementation issues

An important aspect on the application of computer vision techniques in aerial robots is the computational power needed. Also, the application of computer vision techniques on ground computers is seriously constrained by the communication bandwidth and then it cannot be considered for certain applications, such as motion control.

The CMU helicopter uses custom vision hardware to perform on-board all the computer vision and control activities. On-board computer vision is needed for helicopter control (visual odometer) and for efficient robotic tracking of moving objects. Using dedicated hardware, it can operate at 60 Hz, 1/30 s of latency. The system is included into the control loop, and it is used to stabilise and maneuver small helicopters over reasonable speeds (15 mph).

The BEAR project uses off-the-shelf hardware for vision-related processing, consisting on a Little Board PC running Linux, separated from the navigation computer. They can process images at 30 Hz.

A flexible runtime system for image processing has been developed and implemented inside the WITAS UAV platform ([Nordberg, Forsee'n, Wiklund, & Andersson, 2002](#)). The image processing module communicates with the rest of the system using CORBA. Also, important efforts have been devoted in the WITAS project in order to design custom and highly efficient vision algorithms for filtering and feature extraction.

5. Multi-robot systems

Over past few years research on the coordination and cooperation of multiple UAVs and of multiple aerial and ground autonomous systems has been conducted.

Several efforts are related to the coordination of homogeneous teams of aeroplanes ([McLain, 2000](#)). The problems are related to the control of multiple UAVs (aeroplanes) in close-formation flight, as for example in the Air Force Research Laboratory ([Schumacher & Singh, 2000](#)); or in the Air Force Institute of Technology ([Hall & Pachter, 1999](#)). The Phoenix Project at Princeton University also considers the coordinated flight of a fleet of

homogeneous UAVs (aeroplanes) and the design scenario is autonomous aerobatic manoeuvring. The problem of autonomous formation flight control is also considered in [Giulietti, Pollini, and Innocenti \(2000\)](#) where a standard linear quadratic control structure is synthesized for each vehicle and for the formation. The definition of a formation management structure capable of dealing with a variety of transmission and communications failures between aircraft is also presented.

Formation flights have been proposed as a way to deploy multiple sensors on the terrain. This strategy can be considered as biologically inspired (animals that have the ability to form formations such as flocks of birds).

Multiple flying helicopters and groups of helicopters and ground vehicles are considered in the BEAR project. The research includes hierarchical multiagents system architectures for coordinated team efforts, vision-based pose-estimation of multiple UAVs and ground vehicles, and pursuit-evasion games in which a team of UAVs and ground vehicles pursue a second team of evaders while concurrently building a map in an unknown environment ([Vidal et al., 2002](#)).

The cooperation between aerial and ground robots is also researched at USC. In [Sukhatme, Montgomery, and Vaughan \(2001\)](#) different cooperation cases are studied such as the use of an aerial robot in a marsupial-style deployment of a small wheeled robot and the localisation of an aerial robot by visually locating and communicating with a ground robot. "Micro" air vehicles (MAV) are also researched ([Vaughan, Sukhatme, Mesa-Martinez, & Montgomery, 2000](#)) in this framework. Furthermore, the Raptor project ([Saripally, Naffin, & Sukhatme, 2002](#)) considers the use of small electric-powered radio-controlled model helicopters (electric powered) with micro-controllers and Micro Electro-Mechanical based Sensors (MEMS). Relative localization of each robot will be accomplished using only local sensing (CMOS camera), in contrast to global localization techniques (GPS), and then it could be applied in environments where GPS is not available (i.e. indoors or between skyscrapers). Each robot will only have knowledge of its relative location with respect to one or more of its neighbours.

The main objective of the COMETS project ([Merino & Ollero, 2002](#)) is to design and implement a distributed control system for cooperative detection and monitoring using heterogeneous unmanned aerial vehicles (UAVs). Particularly, both helicopters and airships are considered.

In order to achieve this general objective, a control architecture has been designed, new control techniques are being developed, and the integration of distributed sensing techniques and real-time image processing capabilities is considered. [Fig. 11](#) shows a general picture of the COMETS system.

The COMETS project exploits the complementarities of different UAVs in missions where the only way to guarantee the success is the cooperation of several autonomous

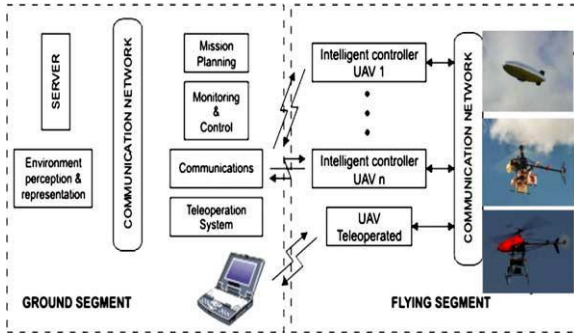


Fig. 11. The COMETS system.

vehicles due to the requirements on the required coverage and the different characteristics of the vehicles. Furthermore, this approach leads to redundant solutions offering greater fault tolerance and flexibility when comparing with the use of a single UAV with long endurance flight and important on-board capabilities. The project also involves the cooperation between robotic aerial vehicles and remotely piloted vehicles. This approach will take benefit from the expertise of human operators in missions where the full autonomy is very difficult to achieve, but pose additional coordination and control problems due to the variability of the human operator.

In order to test and validate these concepts and systems, experiments and demonstrations are being carried out in forest fire alarm confirmation, localisation and monitoring. The first experiments with two helicopters and one airship have been carried out in May 2003.

5.1. Multi-robot perception

Cooperative perception is a key issue in the COMETS project. Several heterogeneous UAVs will be used for mapping, detection and monitoring applications. The UAVs are equipped with sensors of different modalities, such as infrared and visual cameras. The redundancies in the system will be used to reduce false alarms in fire detection activities and for precise localisation of interesting objects. In fire monitoring applications the UAVs will collaborate in order to situate themselves in the best positions to obtain the important characteristics of the fire, avoid smoke, etc.

In Ling, Ridley, Kim, Nettleton, and Sukkarieh (2003) an architecture for multi-vehicle SLAM is studied for its use with UAVs. The issues of data association and communication are dealt with, and some simulation results are presented. In the approach, the map information (represented as the location of a discrete set of landmarks) estimated for each vehicle is propagated to the rest of the fleet. The distributed information is based on the information form of the Kalman filter. Each vehicle uses the information received to update its state and its local map. The fact that the landmarks do not move allows the estimation of other vehicles to be received at arbitrary latency and order.

6. Conclusions

In the last 10 years a significant progress toward autonomous aerial vehicles with on-board intelligent capabilities has been experienced. This progress is fuelling the development of Aerial Robots with significant autonomous capabilities. These systems open new applications in Field Robotics including surveillance, disaster (environmental, industrial and urban) remediation, search and rescue, environment monitoring and many others.

Many different techniques have been applied for UAV control and particularly for autonomous helicopter control. These include techniques to cope with the expertise of human pilots by means of predefined rules or by autonomous learning from the pilots, and model-based control methods. Both linear and nonlinear control techniques have been applied for model-based control. Some of these methods are shortly reviewed in the paper.

Computer vision is the most relevant perception technology applied in aerial robotics perception. It is applied for motion and position estimation, object detection and tracking, autonomous take-off and landing, as well as for applications such as detection, monitoring and terrain mapping.

Finally, the paper has presented a summary of the recent research on multiple aerial robots and coordination of aerial and ground robots, and has shortly presented the COMETS project on the real-time coordination and control of multiple unmanned aerial vehicles.

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