

Numerical modeling and analytical validation of stress intensity factor and crack velocity for SENT tensile specimen of P265GH steel material

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Abstract

During service, most of the mechanical structures are subjected tovariable loads over time, leading to their sudden break and causing serious accidents. The life time of a mechanical structure could be divided into three stages: Stage I; the initiation, stage II; the slow-spreading and stage III; the suddenspreading. The mechanical behavior of parts is directly related to the number of cycles before initiation and slow propagation that could occur in stress concentration zones, or simply because of manufacturing defects.

The aim of this study is to predict the lifetime of Pressurized Structures by modelingnumerically the evolution of the crack propagation velocity through a SENT specimen using the finite element code CASTEM2013.

The results show that the propagation velocity of crack increases by increasing the length of the crack and the stress level.

Key Words:

Notch; Finite element model; Stress; Stress intensity factor; Pressure vessels; number of cycles; Propagation Velocity; Length of Crack.

I. INTRODUCTION

On the microscopic scale, any material has a homogeneities, defect (in inclusions. manufacturing defects, etc.) and anv mechanical part could present section changes or rough surface states. Since these conditions favor the appearance of stress concentrations, we should often consider the possibility of crack initiation as well as its propagation when calculating a structure. For this reason, the designers of structures or any element subjected to cyclic loadings should not only take into account the possibility of cracking, but also estimate the velocity of crack propagation, to ensure that these cracks do not reach the critical length, which will inevitably lead to failure.

The behavior simulation with FEM has been presented by many authors with the aim to improve the knowledge of predicting trends. A. HACHIM [1][2] presented a finite element based approach to simulate a Double Edge Notch Tension specimen of S355 Steel; he studied the behavior of the material in the presence of defects. Y.HIROSHI [3] presented the critical stress intensity factor on SENT specimen. HAKIMI [4] generalized the study of cylindrical and spherical shells under pressure. He proposes an approach based on the stress intensity factors K and J integral



obtained numerically or by simplifying methods for semielliptical cracks internal or external, longitudinal or circumferential. Topper and El Haddad [5] use the stress distribution at the head of a short crack emanating from a notch, to establish the boundary conditions between the stress concentration factor theory of a blunt cut and that an acute notch.

The finite element method adopted in this paper is the most used for real applications to provide a robust solution for most industrial problems.

This work is devoted to the finite element analysis of a tensile specimen simple notched (SENT) using Castem2013[6] numerical code, through which we study the numerical crack velocity to predict the lifetime of the equipment, The material which is the subject of this research is a P265GH steel commonly used in pressure vessels.

II. Experimentation

To extract the mechanical characteristics of the P265GH steel used in our program, tensile tests on standard specimens (Figure1) were conducted in different directions of rolling (longitudinal and transversal). The test curves showing the stress versus strain are given in Figure 2:



Figure 1: Dimensions of the standard test specimen



Figure 2: Stress strain curve test

By comparing the mechanical characteristics of the specimen in both rolling directions, it is found that there is a negligible difference between the two curves. The mechanical characteristics of P265GH steel, at the ambient temperature, are reported in the table1:

Table 1:Mechanical properties of the material

		Breaki		
Young'smodu	elasticlim	ng	Flongati	Poisso
lus	it:	stress:	on %	n's
$E(MPa \Box \Box$	$\square_{e}(\square Pa)$	\Box_{g}	011 /0	ratio v
		$(\Box Pa)$		
2.10^5	320	470	35	0,3

We notice that the elongation is about 35%, which is higher than 14% required by the CODAP [7]. Therefore, this P265GH steel used is well adapted for pressurized structures.

III. Numericalmodeling

The calculation code Cast3m 2013 is used to construct a finite element model in order to analyze SENT specimen behavior subjected to tensile stress. In what follows, we describe FE modeling.

a. Geometry

The geometry and dimensions of the studied specimen are shown in Figure 3. Since the study is restricted to the mode I, The



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specimen was subjected to tensile solicitations.



Figure 3 : The geometry and dimensions of the studied specimen

b. Mesh and boundary conditions

By taking into consideration the symmetry of the problem, only half of the test specimen is discretized. Because the numerical results are intended for analysis of fracture mechanics, special attention is paid to mesh principally in crack and its vicinity (Mesh Refinement using Barsoum elements).[8]. Details of the mesh are illustrated in Figure 4 and Figure 5.



c.Loading

The simulated load is a tensile solicitation along the longitudinal axis of the specimen. To avoid bending or twisting parasite and to ensure that the tensile stress is perfectly aligned; it is applied on the specimen via a rigid triangle indicated by the arrow 4. The selected loads are calibrated in such a way that the applied nominal stresses are respectively 148 MPa, 284MPa and 356MPa.

IV. Results and discussion

a. Evolution of crack velocity according to length of notch

When a crack is formed, its length increases in proportion to the increase of cyclesnumber. The crack velocity is determined by the equation (1):

 $da/dN = C (\Delta K)^{m}(1)$

with:

- da / dN: The crack velocity (where a is the crack length and N is the number of applied cycles)
- ΔK: stress intensity factor [9]
- C and m: constants of materials

According to the equation (1), it is clear that the crack velocity increases constantly until the sudden break of the part.

Cracking is generally described by the following macroscopic three stages:

a) Stage I (propagation velocity less than 10⁻⁵ mm / cycle)

b) Stage II (propagation velocity is between 10^{-5} and 10^{-3} mm / cycle)

c) Stage III (rapid spread leading to sudden break than 10^{-3} mm / cycle).



The curves in Figure 6 illustrate the evolution of crack velocity for the three levels of applied stress: $\Delta \sigma = 148$ MPa, 284MPa, 356MPa



Figure 6 : The crack velocity for the three applied stresses ($\Delta \sigma$ = 356MPa, 284MPa, 148MPa)

• A low level of stress ($\Delta \sigma = 148$ MPa), the crack velocity is practically located between 10^{-5} and $2x10^{-4}$ mm/ cycle, which corresponds mainly to the end of stage I and stage II of crack propagation.

• At a medium level of stress ($\Delta \sigma = 284$ MPa), the crack velocity is practically located between $4x10^{-5}$ and $2x10^{-3}$ mm / cycle, which corresponds to stage II propagation and early stage III.

• A high level of stress ($\Delta \sigma = 356$ MPa), the crack velocity is in the range $1,1 \times 10^{-4}$ and $2,4 \times 10^{-3}$ mm / cycle, the propagation is mainly in stage II and stage III.

The analysis of these results shows that there is a significant increase in crack velocity according to the crack length.

b. Comparison of the crack velocity according to stress levels

The curves in Figure 7 show the comparison of crack velocity, for three levels of applied stresses: $\Delta \sigma = 148$ MPa, 284MPa, 356MPa.



Figure 7 : Comparison of the crack velocity according to stress levels ($\Delta \sigma$ = 356MPa, 284MPa, 148MPa)

The analysis of these results shows that the evolution of the crack velocity accelerates by increasing stress and crack length. But this variation becomes more aggressive by increasing the stress rather than the length of the crack. The sudden crack propagation (stage III) starts for lower notch lengths and high load (7 mm for $\Delta \sigma = 356$ MPa and 9 mm $\Delta \sigma = 284$ MPa)

c. Evolution of the numerical crack velocity depending on the variation of the stress intensity factor

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Figure 8 illustrates the evolution of the numerical velocity logarithm as a function of stress intensity factor for the three applied stresses ($\Delta \sigma = 148$ MPa, 284MPa, 356MPa.)



Figure 8 : Comparison of the crack velocity for three applied stresses

According to Figure 8, it is noticed that:

- Even if the load varies, the representative points of the crack velocity log (da / dN) according to log (ΔK) are located on the same line (linear part), it is regime 2 called regime of Paris.
- The values of $(\log (da / dN), \log (\Delta K))$ increase when the load increases.
- There is a gradual acceleration of the crack velocity when thevalues of the stress intensity factor approach the critical value (K_{IC} value where there is a sudden break of the specimen).

V. Conclusion

A numerical finite element analysis (using the Cast3m2013 code) of SENT specimen loaded in mode I, was carried out for three levels of nominal stress. Results show that the crack velocity varies with the intensity of the applied stress and also according to the

progress of the crack length. For a relatively low nominal stress level about 50% (148 MPa) of the elastic limit, the crack velocity is practically located between 10^{-5} and 1.2×10^{-4} mm / cycle. For a higher stress level about 90% of the elastic limit, the velocity of cracking increases in the interval between 4×10^{-5} and 1.2×10^{-3} mm / cycles. For the highest stress level about 110% of the elastic limit, the crack velocity is practically located between 1.1×10^{-4} et 2.4×10^{-3} mm/cycle mm / cycle

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